



**Preliminary analysis of
"The Vision for Space Exploration" program and
estimation of the role of Russian space firms in
its realization**

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General estimation of "The Vision for Space Exploration" program



Comprehensive examination of the main directions of Universe exploration

Balanced and logical approach to the program realization

Pragmatic approach to formulating of the goals and objectives of the program

Planned involvement of world space community in the process of program realization



Moments, are needed to be taken into account during further program development



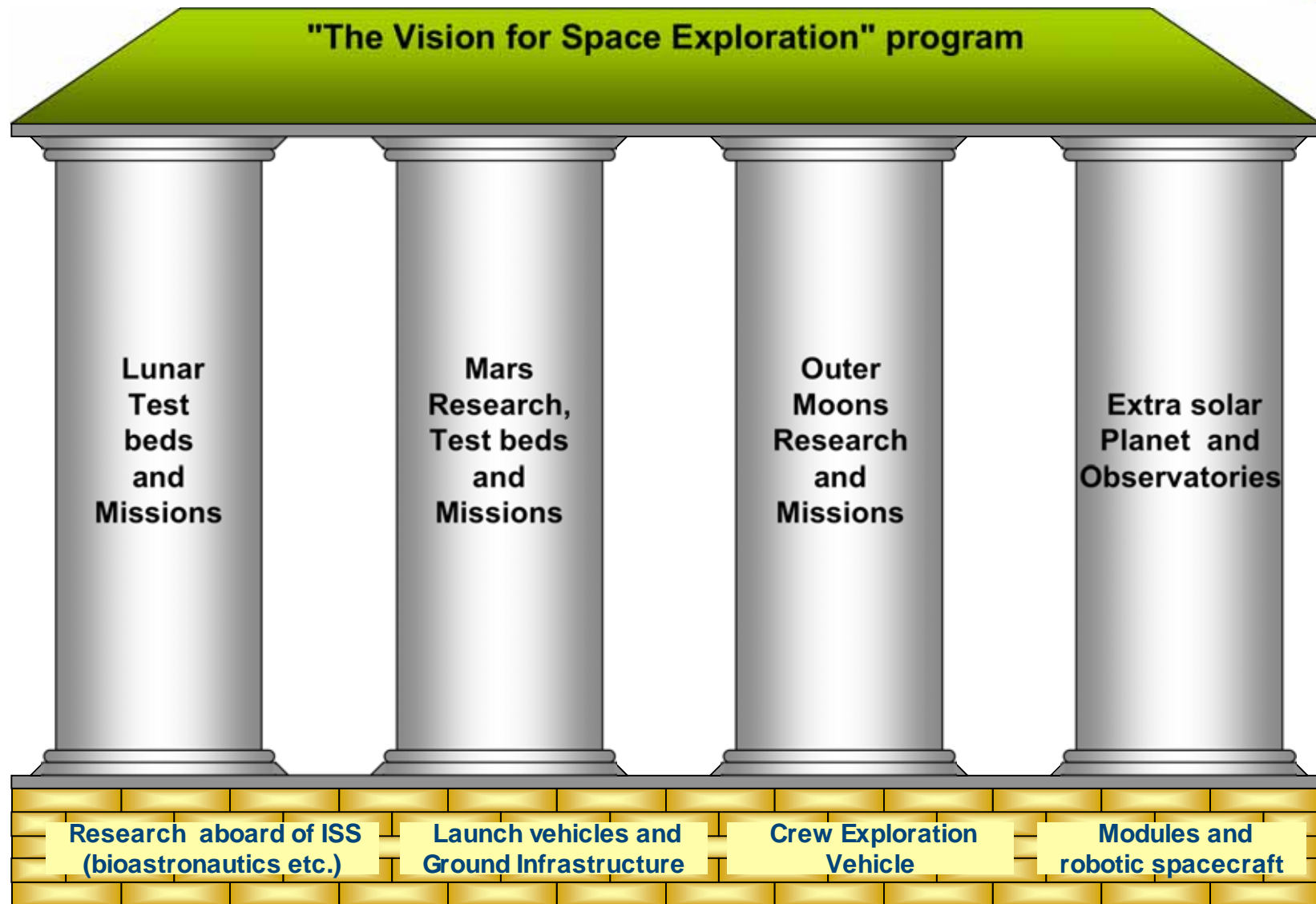
Strategic estimation of the ability of program realization. A creation of optimal planned mission schemes.

A creation of concrete program of development and utilization of space means.

Detailed analysis of technological and scientific "baggage".



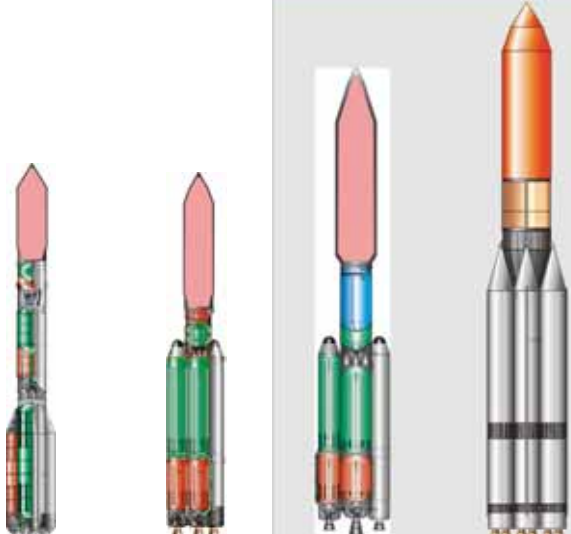




Image model of "The Vision for Space Exploration"



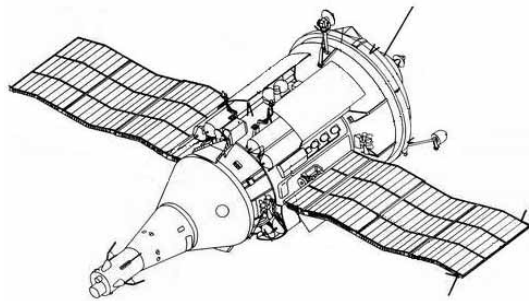
Russian Launch Vehicles



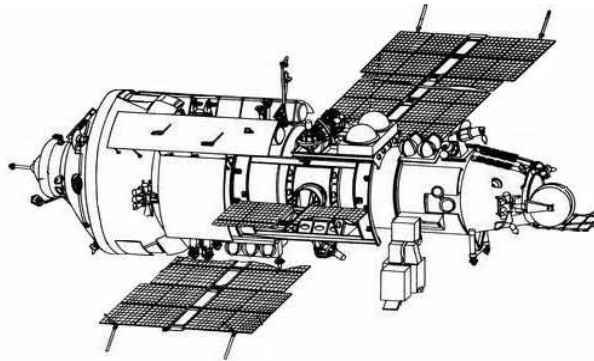
| Developer | Space Launch Vehicles | | |
|-----------------------------|---|---|---|
| | Light class | Medium class | Heavy and super heavy class |
| Khrunichev Space Center |  Rockot Angara-1.1 Angara-1.2 |  Angara-A3 Angara-A3M |  Proton-M Angara-A5 Angara-7 Angara-100 |
| Other Russian Space Firms |  Kosmos Strela Dnepr |  Soyuz-2 | |
| Energy capability to LEO, t | 1 - 4 | 7 - 15 | 22 - 100 |



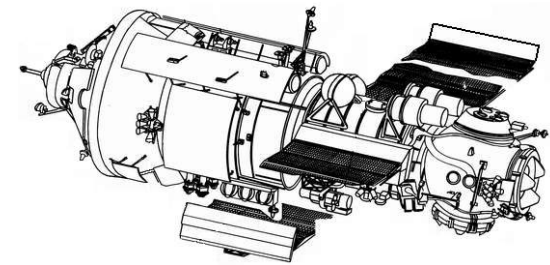
Modules for Manned Space Stations



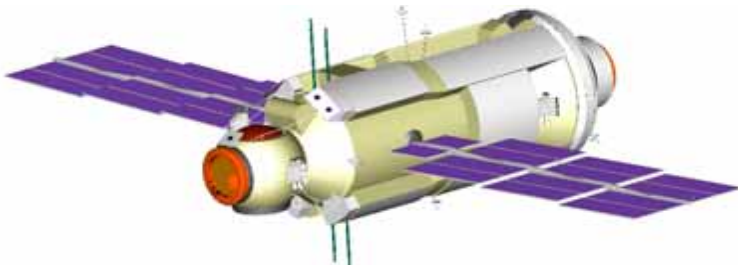
**Cargo Transport Vehicle
for Salyut Orbital Station**



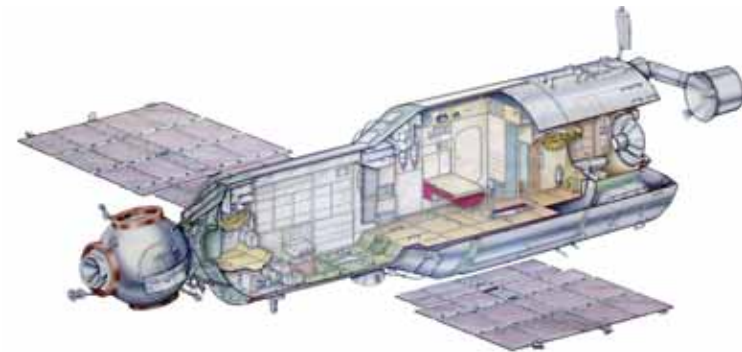
**Kvant-2-type Modules for
Mir Orbital Station**



**Kristall-type Modules for
Mir Orbital Station**



**FGB Zarya Power Supply Module
for International Space Station**



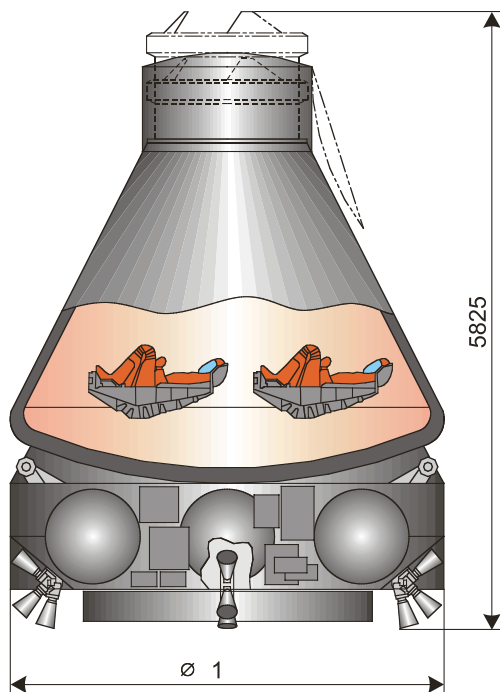
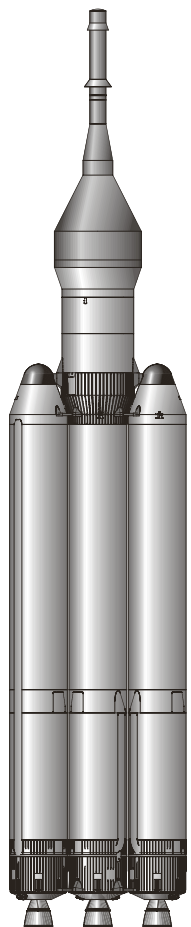
**Basic Modules similar to Zvezda Service
Module for International Space Station**





Khrunichev proposal for Manned Spacecraft

Многоразовый пилотируемый корабль на базе
транспортного корабля снабжения



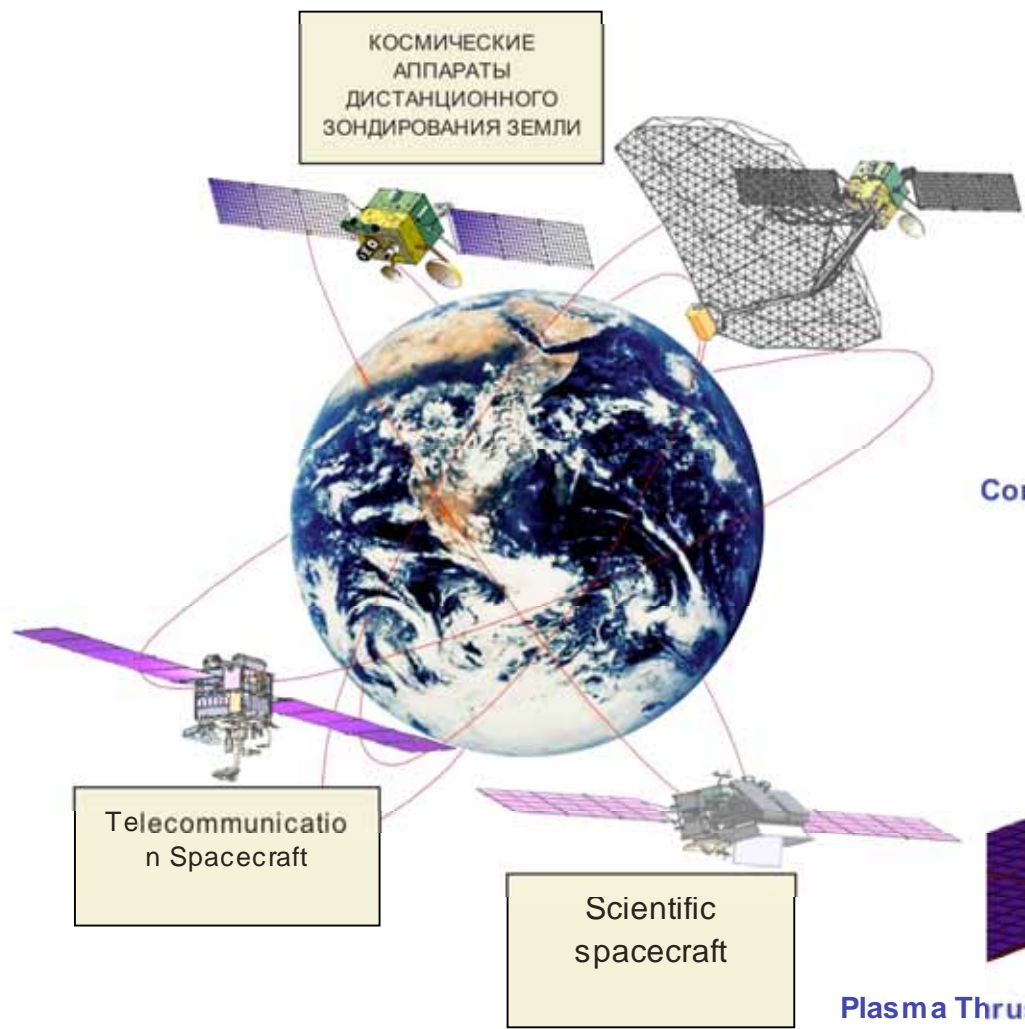
Main Characteristics

| | |
|--|---------|
| Initial mass | 14 t |
| Crew number | 2- 6 |
| Calculated number of flights | 10 |
| Payload mass to LEO (cargo option) | 6350 kg |
| Payload mass returned from LEO (cargo option) | 1870 кг |





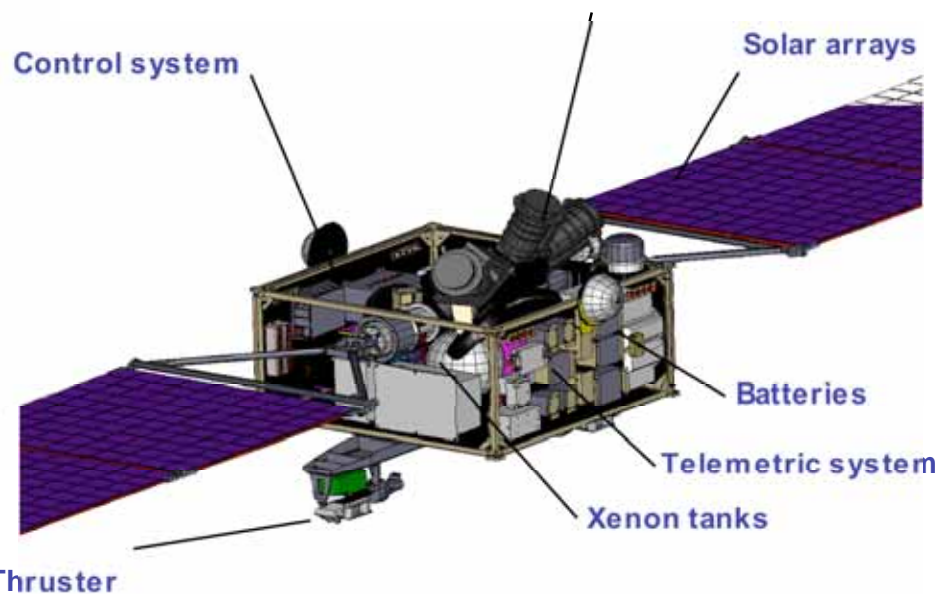
Unified Space Bus "Yacht" and Spacecraft on its basis



Main characteristics of Unified Space Bus "Yacht"

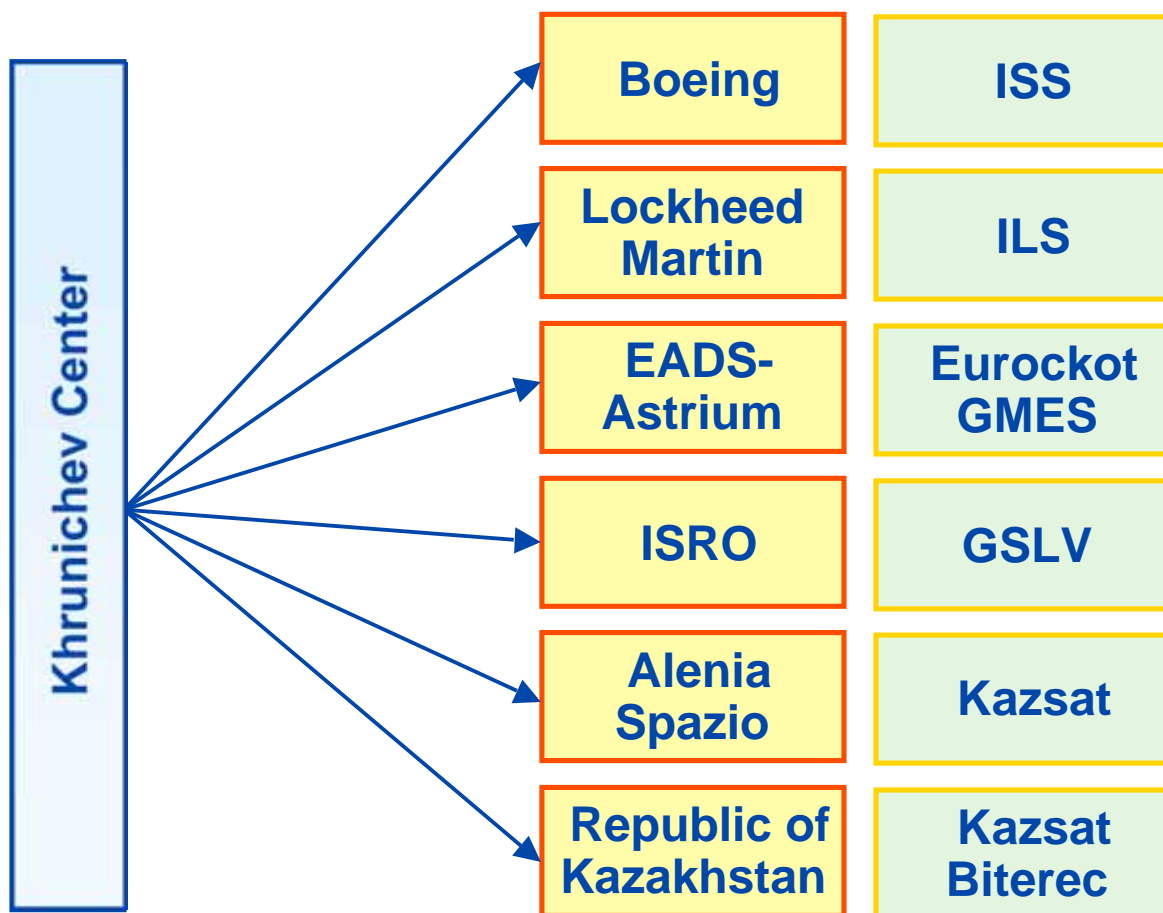
| | |
|--|-------------|
| Bus mass, kg | 350...500 |
| Main dimensions, m | 1,2x1,2x0,6 |
| Solar arrays power, k Wt | 1,5...3,9 |
| Active life time, years | up to 12 |
| Calculated reliability during active life time | 0,95 |

Astro navigational system





Khronichev Participation in International Programs



Interaction and Management Structure

